# **Development of Hybrid Floor Isolation System** with Semi-active Control

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#### SUMMARY

The floor isolation system is designed for one floor of the structure. Limited by the space, relatively small seismic clearance is preferred. However, the long period type input motion will cause large displacement and even pounding. To design a floor isolation system which works effectively under both short and long period motions, the semi-active technology is employed. This paper presents the shaking table test results of a semi-active controlled floor isolation system equipped with a MR damper. The LQR control algorithm is used to calculate the desired force and a PI force tracking scheme is designed to track the design force. A new method is proposed to update the control gain based on the frequency components of the input motion. The test results show that the semi-active controlled floor isolation system can effectively reduce the acceleration while maintaining the displacement in an acceptable level.

Keywords: hybrid floor isolation system, semi-active control, shaking table test

## **1. INTRODUCTION**

Base isolation is one of the most successful and widely-applied techniques of mitigating structural vibration during earthquakes. It is also designed to protect the important equipment so as to maintain the functionality of the structure during and immediately after the earthquake (for example, Sato et al. 2011). However, the base isolation technique applied to the entire building may not be practical in some cases, including those for retrofit. In such a situation, the floor isolation system (FIS) that is designed for one floor of the structure is a cost-effective alternative. The isolation system with passive damping is found not necessarily optimal for a wide band of frequencies and intensities of the input motions (Nagrajaiah and Sun 2000). Semi-active control is another alternative to augment the control force. Recent research proves that a semi-active controlled base isolation system performs better than an equivalent passive controlled system (Yoshioka et al. 2002).

This paper experimentally investigated a semi-active controlled hybrid floor isolation system (HFIS) employing a MR fluid damper for the protection of contents inside of the room. A series of preliminary tests were performed to examine the dynamic behavior of the MR damper. Next, a PI controller was built to calculate the command current corresponding to the design forces obtained from a LQR controller. A series of shaking table tests were performed using the control strategy proposed. Passive control was also tested for comparison with the semi-active control.

# 2. INVERSE DYNAMIC MODEL OF MR DAMPER

A MR damper is essentially a nonlinear device, and it is widely used in semi-active control. To develop control algorithms that take the maximum advantage of the unique features of the MR damper, a model must be established to describe the relationship between the input current and the output force. In real applications, the output force is the known parameter calculated from the adopted control



algorithm, and the input current is unknown. Various kinds of models have been developed to model the characteristics of MR dampers (Wang and Liao 2011). However, the inverse dynamic model for MR dampers, which calculates the current based on the desired force, is still immature. The dynamic model, such as the phenomenological model (Spencer et al. 1997), can effectively emulate the dynamic behavior of the MR damper. Due to its complexity, however, the command signals cannot be calculated explicitly. Instead of establishing an inverse dynamic model, a clipped optimal control was often used (Dyke et al, 1996). The disadvantage of this control is that sometimes the sudden change in voltage would cause increases of the system response. To solve this problem, an inverse dynamic model, so called PI controller, was designed in this study. The test program conducted to evaluate the MR damper characteristics can be found in Shi et al. 2012.

#### 2.1. Transfer Function of the MR Damper

A triangular type motion was used to drive the MR damper at a constant velocity. The input current to the MR damper was a sinusoidal signal, and a wide range of frequencies and amplitudes were chosen to obtain the transfer function from the current to output force. Fig. 2.1 shows the bode diagram of the transfer function. The magnitude is different for different currents and the magnitude changes in accordance to the frequency rather than remains at a constant value. Observed also in the figure (Phase) is that there is a time lag from the input current to the output force. A one order transfer function was used to fit the test results with the testing conditions of 1A, 40 mm/s as shown in Eqn. 2.1



Figure 2.1. Transfer Function of MR Damper

#### 2.2. MR Damper Models

A Bingham model (see Fig. 2.2), which is essentially a bilinear model, was adopted to describe the MR damper behavior in Eqn. 2.2.

$$u = \operatorname{sign}(\dot{x}) f_{c} + c_{0} \dot{x} + f_{0}$$
(2.2)

where *u* is the control force determined from the LQ control algorithm. Test results (Shi et al. 2012) show that the parameter  $f_c$  has a linear relationship with current *I* as shown in Fig. 2.3, and  $f_c = 2.564 I + 0.55$  (kN);  $c_0=1.060$  (kN.s/m), and  $f_0=0$ . Note that the data were obtained from the test with constant current. The magnitude changing of the transfer function is not included in this method. Since the relation between the current and the output force is linear, it is convenient to acquire current from its inverse dynamic model. The inverse model of MR damper can be expressed as:

$$I = (u - 0.55 - 1.06\dot{x}) / (2.564 \text{sign}(\dot{x}))$$
(2.3)



Figure 2.2. Bingham model

**Figure 2.3.** Relationship between  $f_c$  and I

The main disadvantage of this model is that it does not reproduce the hysteretic behavior observed at a low frequency zone. Furthermore, it cannot consider the non-linear behavior at high frequency regions. In order to compensate the magnitude change of the output force at the high frequency range, a PI controller was designed using the transfer function from the current to output force shown in Eqn. 2.1. Fig. 2.4 shows the block diagram of a PI controller for the MR damper. The PI controller can be expressed in Eqn. 2.4 by employing a high-pass filter to eliminate the influence of low frequency component on the integration part of the PI controller. It is worth to note that this PI controller is designed based on the transfer function obtained with a constant velocity (40 mm/s). The results shown later will prove that it works reasonably for a wide band of velocities thanks to its robustness.

$$I = \left(\frac{s}{s + \omega_{\rm l}}\right) \left(K_{\rm p} + \frac{K_{\rm i}}{s}\right) e \tag{2.4}$$

where  $\omega_1$  is the cut off frequency=0.1Hz;  $K_p$ =0.0002 and  $K_i$ =0.05 were determined by numerical iterations. Fig. 2.5 shows the transfer functions from  $f_{des}$  to  $f_{mea}$ , and from *e* to  $f_{mea}$ . As a comparison, a PI controller with  $K_p$ =0.005 and  $K_i$ =0.05 was also designed. Although the PI controller with  $K_p$ =0.005 was more effective than the controller with  $K_p$ =0.0002, in reproducing the design force at the high frequency region (greater than 10 Hz), the magnitude of the transfer function from *e* to  $f_{mea}$  has a much larger value than that of the system with  $K_p$ =0.0002 when frequency is larger than 1 Hz. In the real test, this may cause large errors due to noises contained in the high frequency region.



Figure 2.4. Block diagram of PI controller

Fig. 2.6 shows the comparison of accuracies for three models: the Bingham model, PI controller and clipped optimal control. The MR damper moves at a constant velocity of 40 mm/s. The comparison shows that the PI controller (with  $K_p$ =0.0002 and  $K_i$ =0.05) has smallest errors among the three (there is no data for clipped optimal control at 0.3 and 10 Hz) for frequencies smaller than 5 Hz. In addition, the results show that the PI controller with  $K_p$ =0.005 has larger error than the one with  $K_p$ =0.0002.



**Figure 2.5.** Transfer functions: a) from  $f_{des}$  to  $f_{mea}$ , b) from e to  $f_{mea}$ 



**Figure 2.6.** Comparison of accuracies: a) design force; b) Bingham model; c) PI controller ( $K_p$ =0.0002); d) PI controller ( $K_p$ =0.0002); e) Clipped optimal

## **3. TEST SETUP**

The HFIS proposed in this paper was designed for a four-story RC building served as a hospital, which was tested in E-defense (Sato et al. 2011). It was supposed to be placed on the roof of the hospital. A single degree mass system was used to represent this FI in the test (see Fig. 3.1). Since the mass of the FIS is expected to change in reality by moving in or out the equipment, three mass systems, 35 kN, 62.5 kN and 82.5 kN, were used to simulate such a situation. The mass was supported by four sets of rolling pendulum devices. The natural period created was 3.1 sec. The friction coefficient at the pendulum surface was 1.2%. A MR fluid damper was attached between the mass and shaking table. A magnetostrictive displacement transducer was used to measure the displacement of the piston-rod of the MR damper. The measured displacement and the velocity calculated by the differentiation of the displacement were used as the feedback signal in the control. A load cell was fixed at the end of the MR damper to measure the reaction force. The MR damper mass replaced with an oil damper in the passive control test conducted for comparison. The oil damper provided resistance that was related linearly to the velocity with a damping coefficient of 10 kN.s/m, and its resisting capacity was 10 kN. The damping ratio for the three different mass systems were 70% (35 kN), 39% (62.5 kN) and 29% (82.5 kN), respectively.



Figure 3.1. Specimen and control setup

## 4. CONTROL STRATEGY

4.1. LQR Controller

The FIS was modeled by a single degree of freedom (SDOF) model. The dynamic equation of motion of the model can be expressed as in Eqn. 4.1.

$$\ddot{x} + 2\xi\omega\dot{x} + \omega^2 x + f / m + u / m = -\ddot{x}_g$$

$$\tag{4.1}$$

where x and  $\dot{x}$  are the relative displacement and relative velocity; h,  $\omega$ , and m are the damping ratio, natural frequency, and mass of the FIS; f is the friction of the rolling pendulum system; u is the control force of the MR damper; and  $\ddot{x}_g$  is ground motion. The semi-active controller was designed by combining the friction force and MR damper control force, i.e.  $u^* = u + f$  as the optimal force. The semi-active control for the FIS consisted of two controllers. The LQR algorithm, which has been extensively used for both active and semi-active controls of structures, was selected as the primary controller. Assuming independence of the ground excitation and measurement noises, the absolute acceleration, relative displacement, and control force were weighted using the cost function in Eqn. 4.2.

$$J = \int_0^t \left( \alpha \left( \ddot{x} + \ddot{x}_g \right)^2 + \beta x^2 + \gamma u^{*2} \right) dt = \int_0^t \left( z^T Q z + 2 z^T S u^* + u^{*T} R u^* \right) dt$$
(4.2)

where  $\alpha$ ,  $\beta$ , and  $\gamma$  are weighting coefficients and

$$z = (x \dot{x})^{\mathrm{T}}, Q = \begin{bmatrix} \alpha \omega^4 + \beta & 2\alpha \xi \omega^3 \\ 2\alpha \xi \omega^3 & 4\alpha \xi^2 \omega^2 \end{bmatrix}, S = \begin{bmatrix} \alpha \omega^2 / m \\ 2\alpha \xi \omega / m \end{bmatrix}, R = \alpha / m^2 + r$$
(4.3)

By minimizing Eqn. 4.2, the optimal control force  $u^{*}(t)$  can be obtained from the following equation:

$$u^* = -G_{\mathcal{Z}} \tag{4.4}$$

where G is the feedback gain and is proportional to the mass. Since the MR damper is an intrinsically energy dissipation device, it cannot add mechanical energy to the controlled system. Besides, the capacity of the device has an upper and lower bounds. These constraints form the secondary controller:

$$u^{*} = \begin{cases} u^{*} & (u^{*} \cdot \dot{x} < 0 \& u^{*} < u^{*}_{\max}) \\ u^{*}_{\max} & (u^{*} \cdot \dot{x} < 0 \& u^{*} \ge u^{*}_{\max}) \\ u^{*}_{\min} & (u^{*} \cdot \dot{x} > 0) \end{cases}$$
(4.5)

where  $u_{\min}^*$  and  $u_{\max}^*$  are the minimum and maximum optimal forces provided by the friction and MR damper forces. For simplicity, the output force of semi-active control device was determined by Eqn. 4.6.

$$u = u^* - f$$
 (4.6)

#### 4.2. Control Gain

Calculation of the feedback gain depends on the selection of weighting parameters. The MATLAB simulation results show that the parameters can be set as follows: To best reduce the acceleration, while avoiding an extreme displacement,  $\beta = 10^9$ ,  $\gamma = 0.001$ , and weighting parameter  $\alpha$  has a relationship with the dominant frequency of the input motion as:

$$\alpha = 10^{1.30 + 7.8} \tag{4.7}$$

where  $\omega$  is the dominant frequency of the input motion. Two types of semi-active tests were performed. The first one was performed by assuming that  $\alpha$  was known, and the second one considering the actual situation was performed by assuming that  $\alpha$  was unknown. The control gain was updated by detecting the frequency of the input motion in a 2.5 sec window. The frequency in the first 2.5 sec window was used to update  $\alpha$ , and the calculated gain was used in the next 2.5 sec window. Note that it is not a real time control.

The input motions used in the test were the roof responses of the four-story RC building tested in E-defense (Sato et al. 2011). The first motion JMABF was a short period type motion, which was obtained from the response of the tested structure subjected to a near fault ground motion of 1995 JMA Kobe. The peak acceleration was 10 m/s<sup>2</sup>. Its dominant frequency was 1.4 Hz. The second motion SANBF was a long period motion, which was the roof response of the tested structure subjected to a synthesized ground motion named Sannomaru. The peak acceleration was 3 m/s<sup>2</sup>. Because the dominant low frequency components of the ground motion Sannomaru were not filtered by the structure, the roof response was still dominated at 0.35 Hz. The weighting parameter for the acceleration was set to be  $0.9\alpha$  to avoid large displacements in this long period ground motion.

The short period motion often excites the structure within a short time with great power, while the long period motion shakes the structure for a long time. It is relatively easier to update the control gain based on the frequency information obtained from the long period motions. In such a condition, it is recommendable that the initial value of  $\alpha$  be set at a relatively small value to prevent possible large acceleration responses under short period motions. For the particular case in this study, since the dominant frequency of the motion under short period ground motion was close to the frequency of the first mode of the superstructure, i.e. 1.4 Hz,  $\alpha$  can be set to 9.6 (see Eqn. 4.7) at the initial stage.

# 5. HYBRID FLOOR ISOLATION TEST

Fig. 5.1 shows the test results for the three different mass systems with passive control and two types of semi-active control.



Figure 5.1. Shaking table test results for three control strategies

Fig. 5.1 indicates that the semi-active control could maintain similar control results for different mass systems thanks to its ability to adjust the gain which was proportional to the mass. The response of the FI with passive control could not always stay at an optimal state for different mass systems (different damping). Under JMABF input motion, the acceleration in the semi active control case was reduced

by 33%-73% when compared with the passive control case while both of the two controls remained very close displacement responses; Under SANBF input motion, the acceleration of both the passive and semi-active control for the 62.5 kN system remained similar, but the displacement in the semi-active case was reduced by 23% when compared with the passive control case. Pounding occurred for the passive control and was avoided for semi-active control in the 82.5 kN system. Although in the 35 kN system the passive control reduced the displacement by 18% compared with the semi-active control and had similar acceleration response with semi-active control, but at the expense of a super high damping ratio of 70%.

The semi-active control was able to automatically detect the frequency of the input motion and consequently adjusted the control gain (see the data for 62.5 kN system). The test results for the semi-active control with the updated gains gave similar results compared with the semi-active control with a pre-determined gain, which is not realistic in the actual application.

Fig. 5.2 shows the current calculated by the PI controller, comparison between the design force by LQR algorithm and the actual measured force. The relative velocity is also included in the figure. The results show that the PI controller could track the design force with great accuracy, especilly for SANBF gound motion. The error at the maximum control force point and the RMS error were 1000 N (with maximum control force of 6100 N) and 278 N, respectively, under JMABF input motion; and they were 402 N (with maximum control force of 6901 N) and 277 N, respectively, under SANBF input motion. The maximum velocities under those two motions were 1.04 m/s and 0.4 m/s, respectively. The results show that the PI controller designed based on the results with velocity of 0.04 m/s was very robust.



Figure. 5.2. Applied current and the comparison of design and real measured forces

## 6. CONCLUSIONS

A floor isolation system with semi-active control was developed to protect important equipment and facilities. A series of shaking table tests was performed to validate the properties of the adopted MR damper, and the control strategies were developed. The inverse dynamic model developed was proven to be effective which could track the design force accurately. The floor isolation was tested under passive and semi-active control conditions. The test results show that the acceleration was 33%-73% smaller in the semi-active control case than in the passive control case under the short period JMABF ground motion, while both the two controls maintained similar displacement response; the displacement was 23% smaller in the semi-active control case than in the passive control case while maintaining similar acceleration response under SANBF long period ground motions for a 62.5 kN system. And pounding was avoided in the semi-active control. An algorithm for auto gains was

designed, in which the frequencies characteristics of the input motion were detected and based on the dominate frequency the control gain was updated automatically. The test results for the semi-active control with the auto gains gave similar results compared with the semi-active control with a pre-determined gain.

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